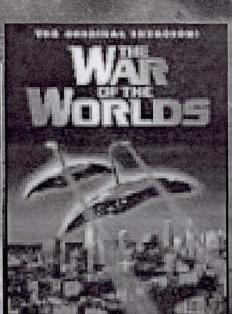
The Future of Human Exploration

Doug Cooke
Manager- Advanced Development Office
NASA – Johnson Space Center
9/07/01

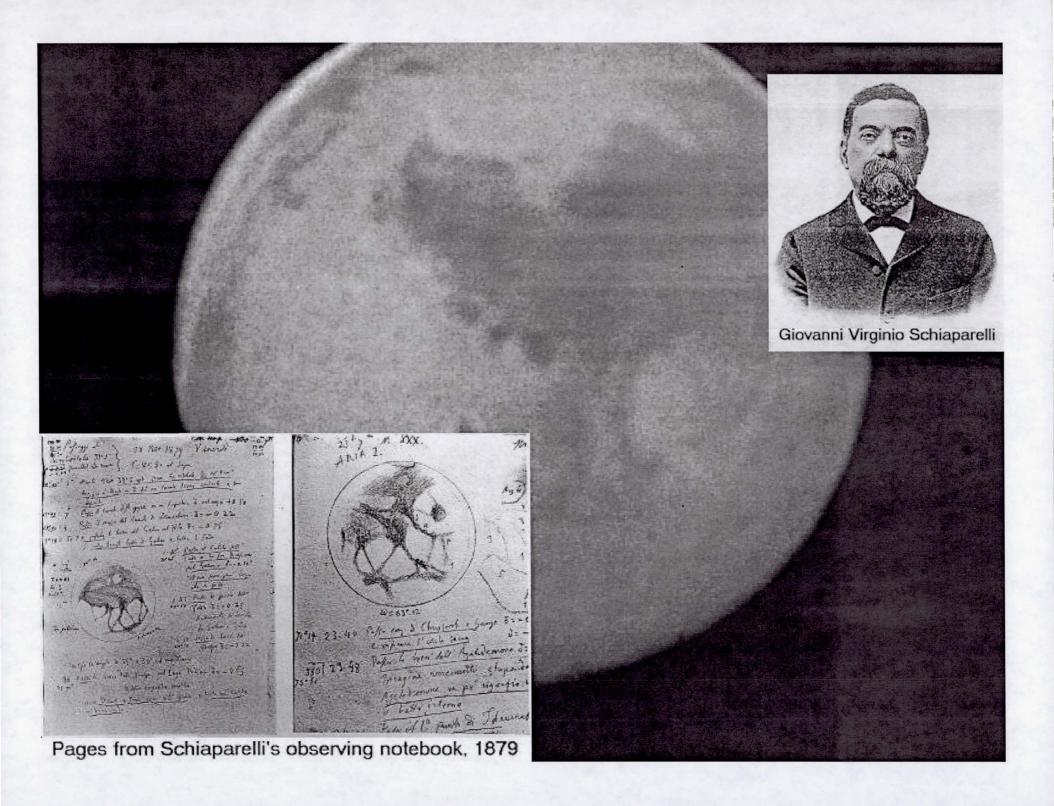


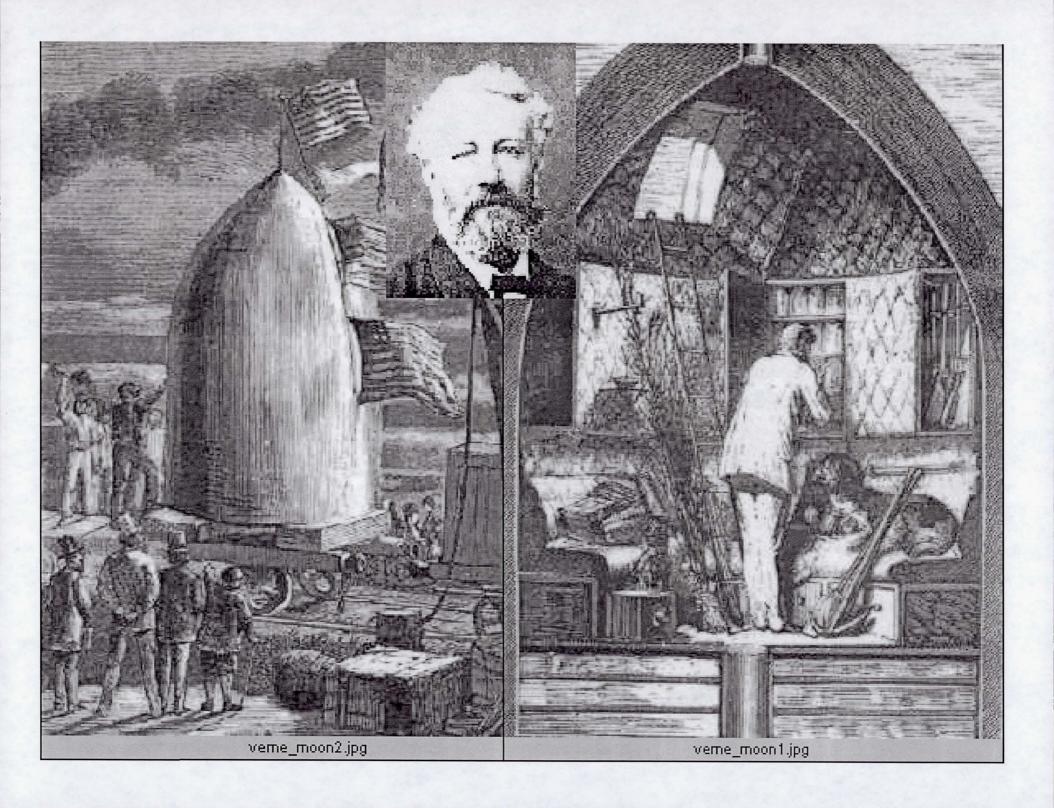


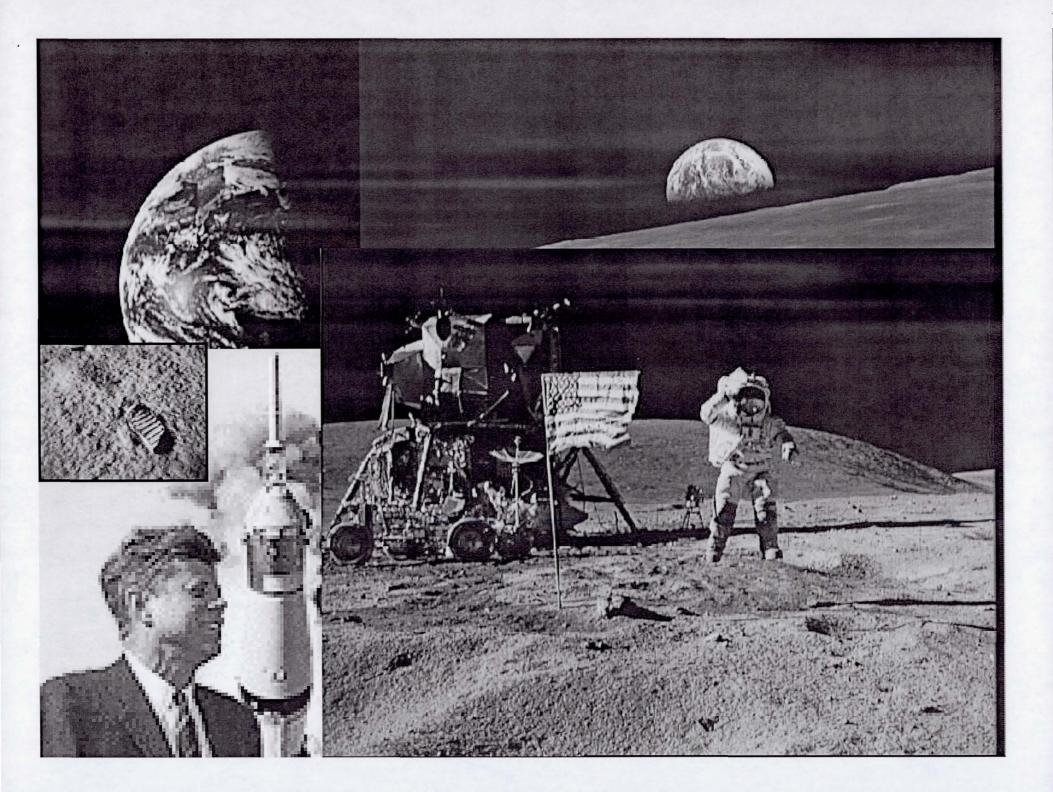




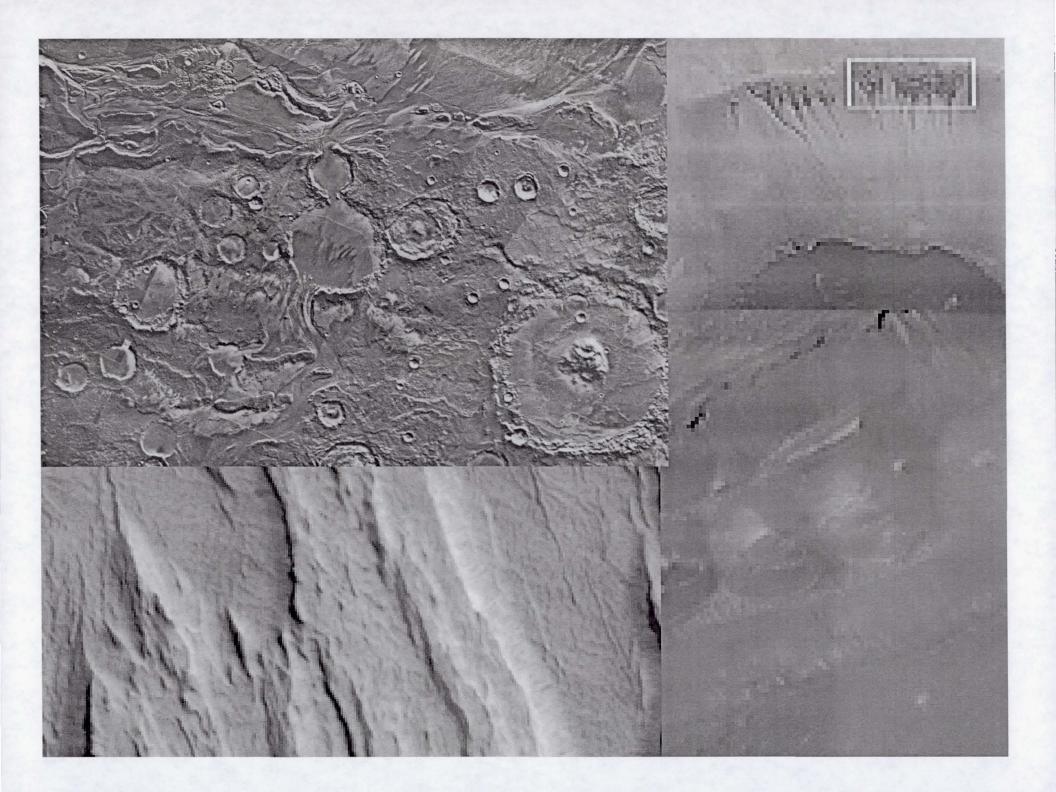


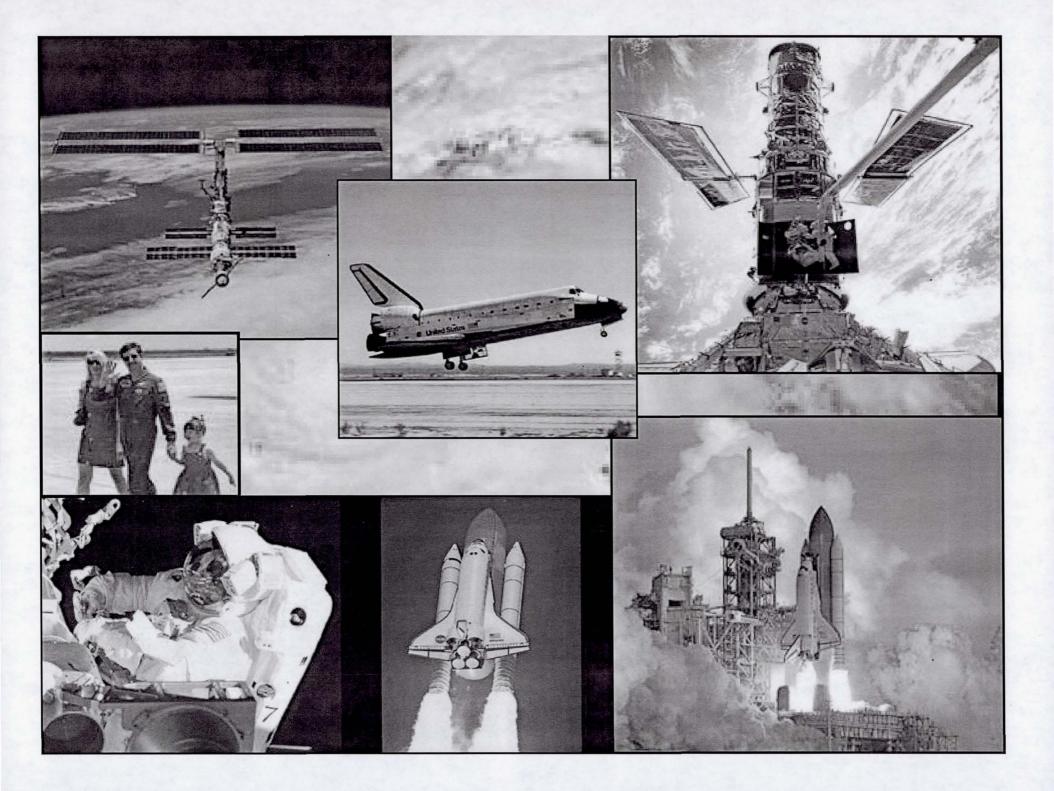




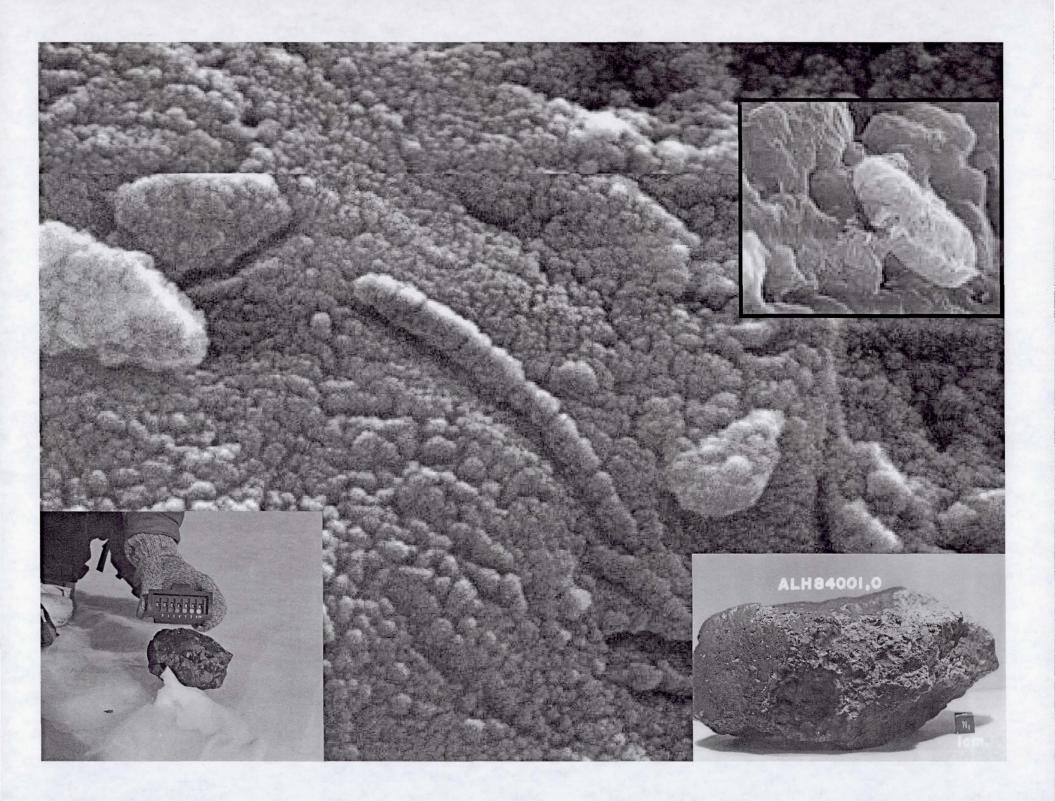




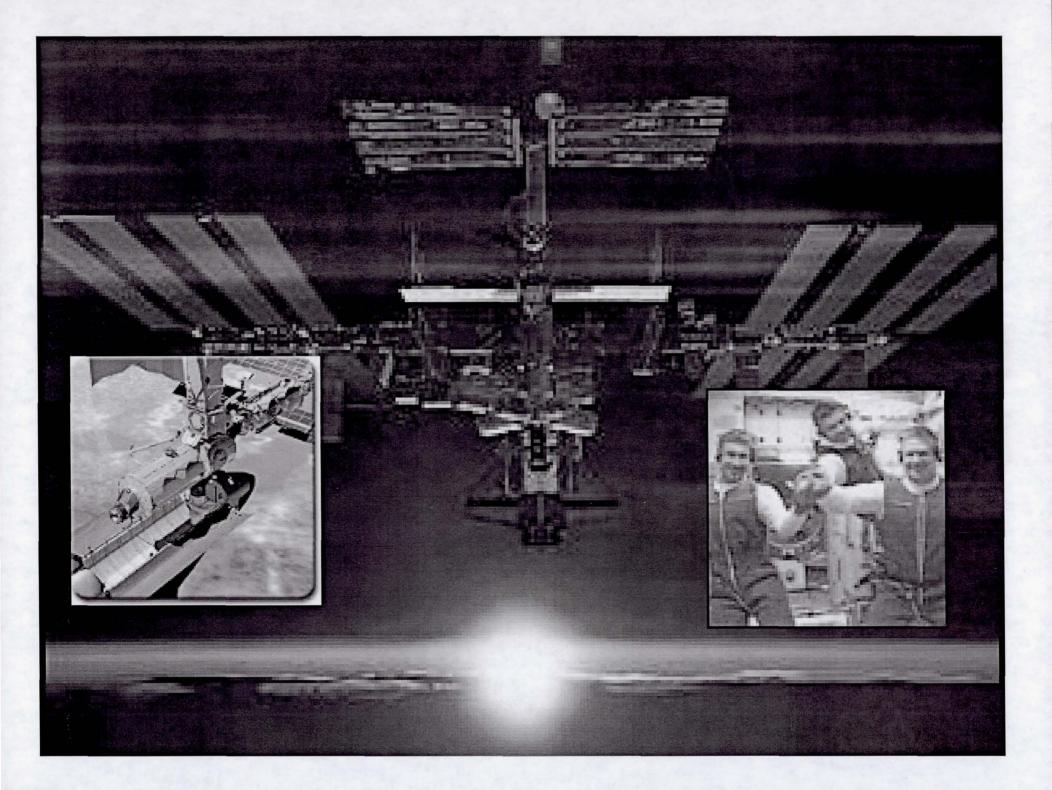












Expanding Knowledge

Science of Earth's Systems

Search for Past Life

Search for Present Life

Origin of Solar System

Mars Climatic History

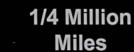
2000 Days?



90 Days



40 Million Miles 400 Million Miles



Developing Operational Capabilities

Operational Safety

Reduced Cost and Risk

Reduced Transit Time

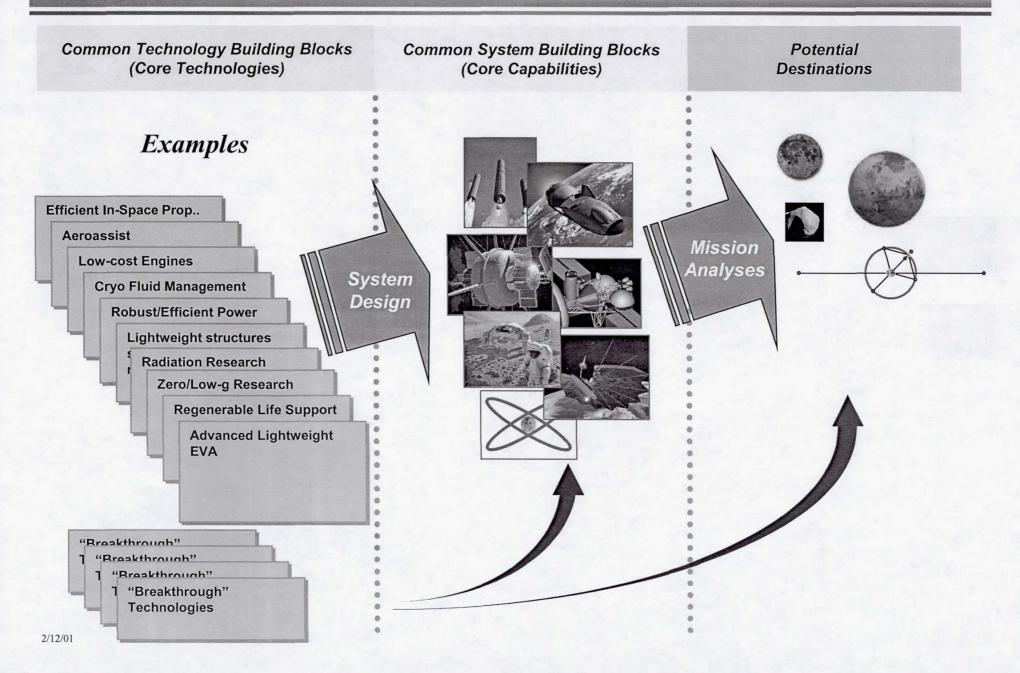
Self Sustainability

Commercial Opportunities



Core Capabilities & Technologies

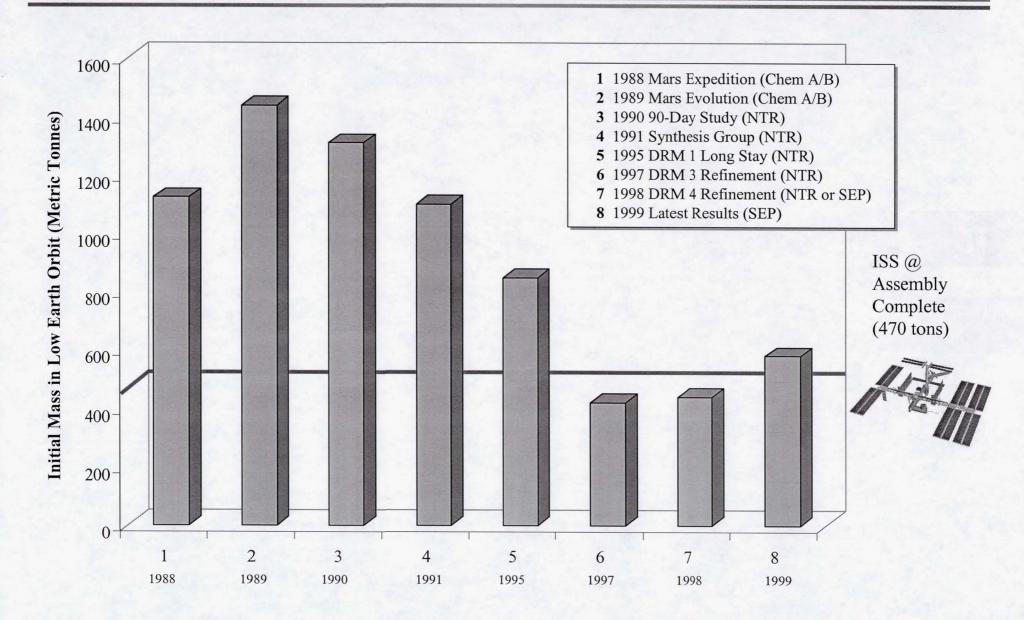






Mars Architecture Mass Comparison

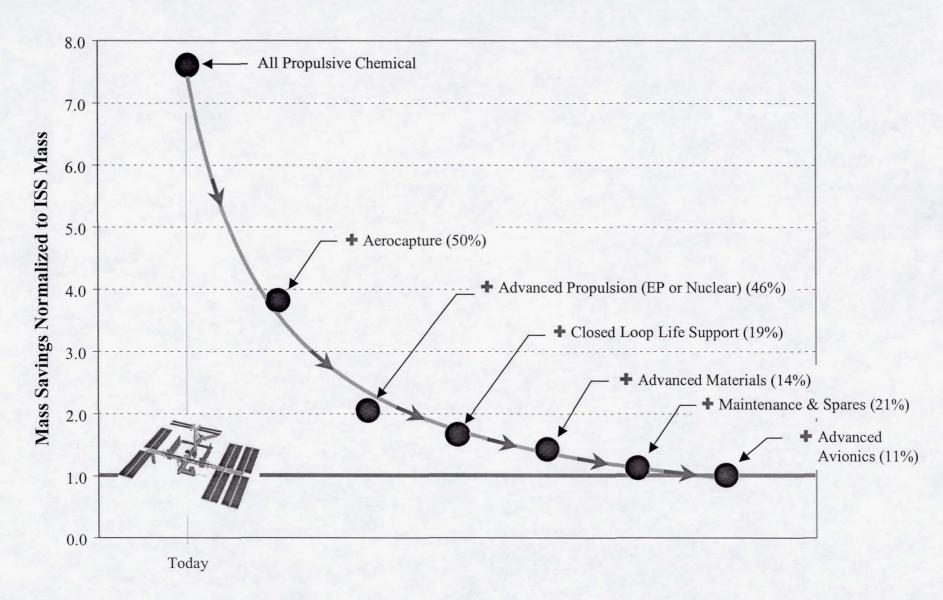






The Value of Technology Investments Mars Mission Example

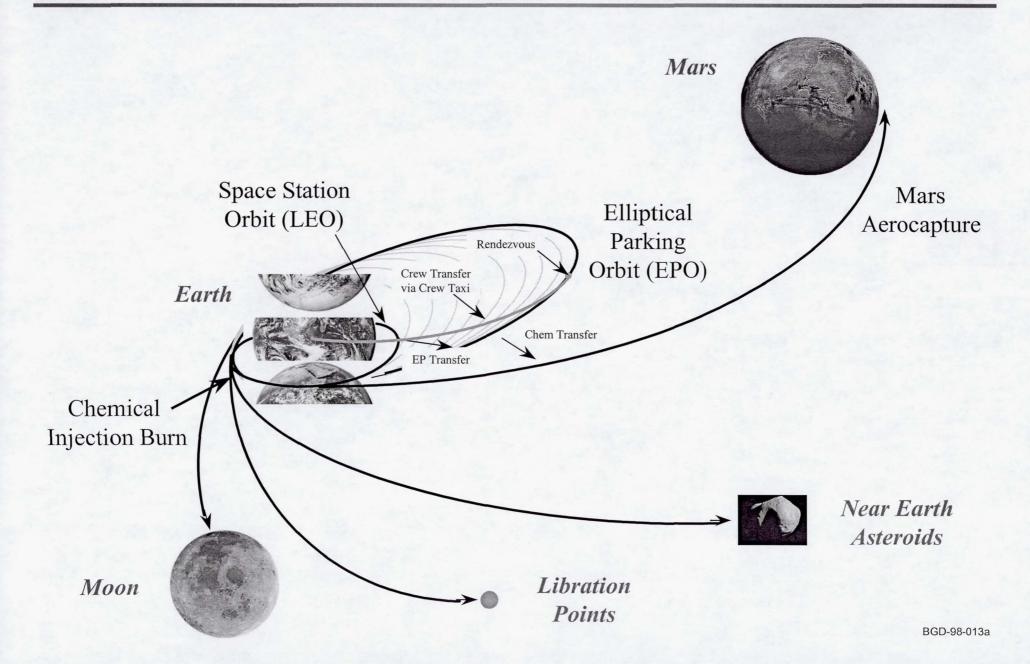






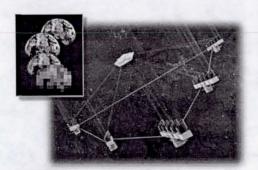
High Earth Orbit Staging Mission Scenarios

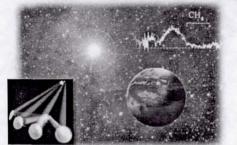














Human Mars Exploration

- · Technology Development
- · Deep-Space Operational Experience

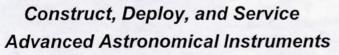
Capabilities

Mission Staging (Hybrid Prop Module Fuel Depot)

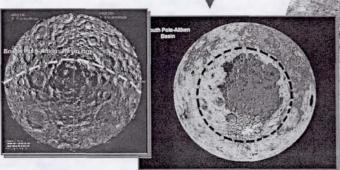




- · Search for Location and Mechanism of Solar Flares
- · Increase Lead Time and Accuracy for Geospace Forecasts



- · Detect Biological Activity on Extra-Solar Planets
- · Image Surfaces of Extra-Solar Planets



Lunar Science

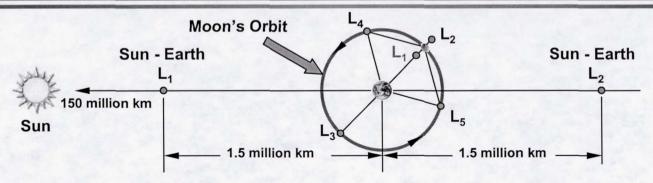
- Impact History in Near-Earth Space
- · Composition of Lunar Mantle
- · Past and Current Solar Activity
- · Poles History of Volatiles in Solar System

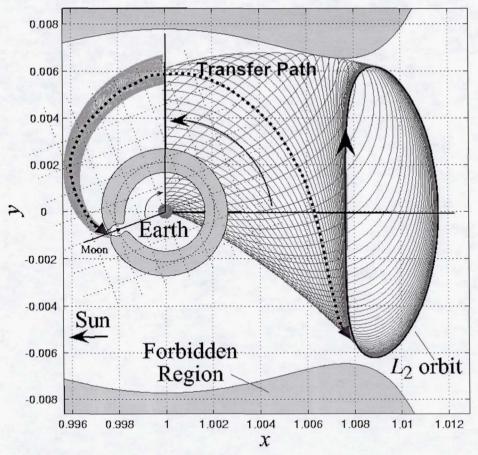




Unique Orbital Dynamics





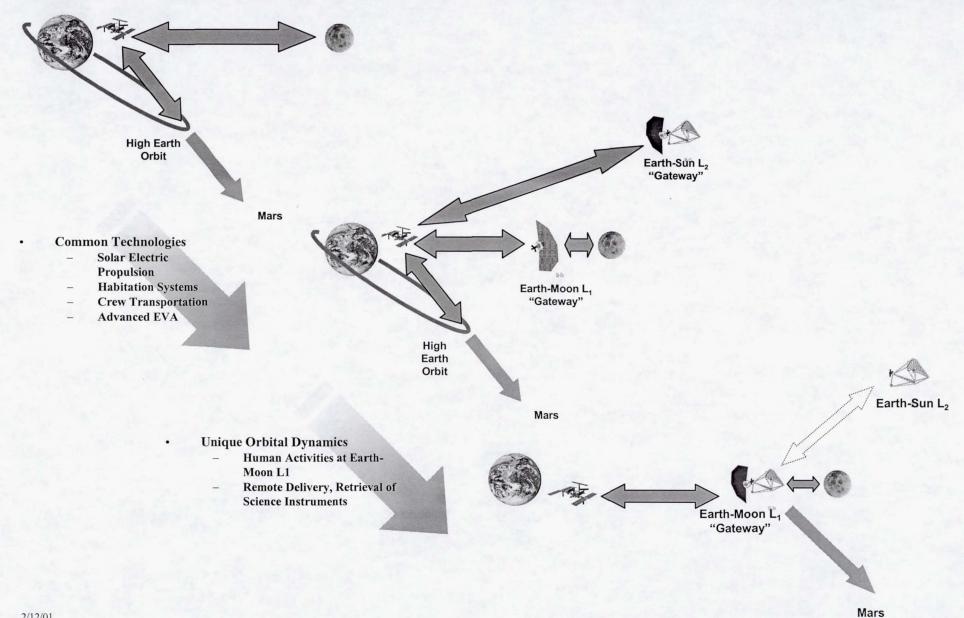


- Orbital Dynamics in Earth-Moon System Leads to Unique Capabilities
 - Low-Energy Transfer from Earth-Moon
 L1 to Solar Libration Points and Return
 - Potential Staging Point for Human Mars Missions
- Allows for Earth-Moon L1
 Deployment and Servicing of Science Assets



Architecture Evolution





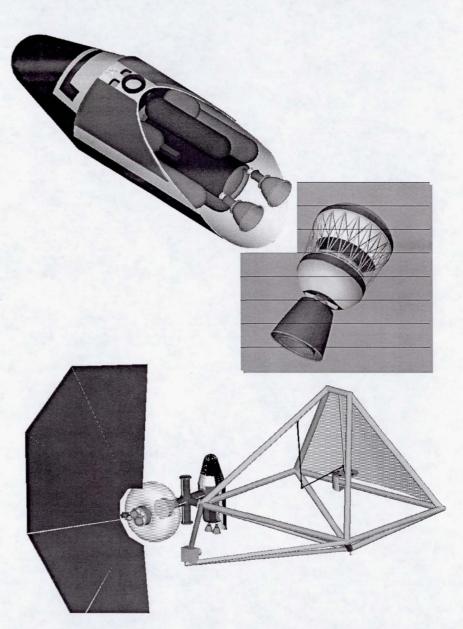


Earth's Neighborhood, Libration Points Architecture



Key Attributes

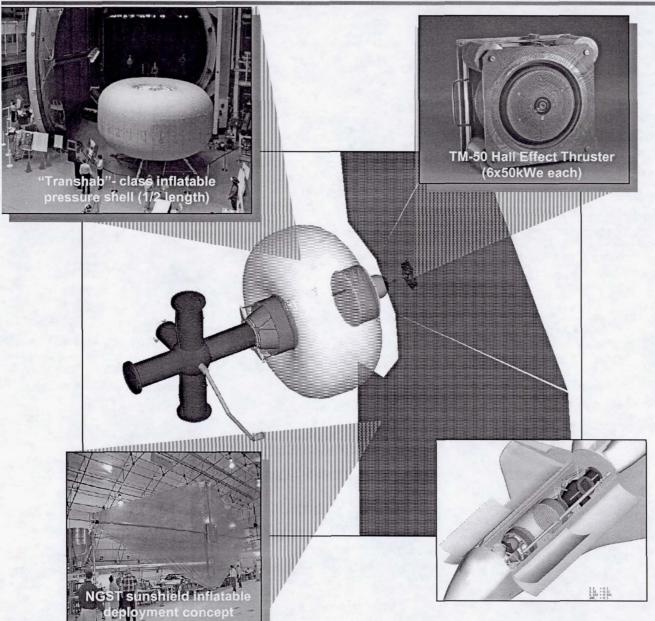
- Crew of 4
- 65 day mission duration
- Deployment, assembly, and servicing of large complex science facilities to achieve revolutionary discoveries
- ISS integral as a staging platform
- Serves as a "stepping stone" by providing an opportunity to test technology and operational concepts, reducing risk of future exploration endeavors
- Architecture can be bought "by the yard" resulting in increasing capabilities and operational experience
- Modest augmentation of commercial launch vehicles
- Common architecture elements for all Earth's Neighborhood missions





Lunar L₁ "Gateway"





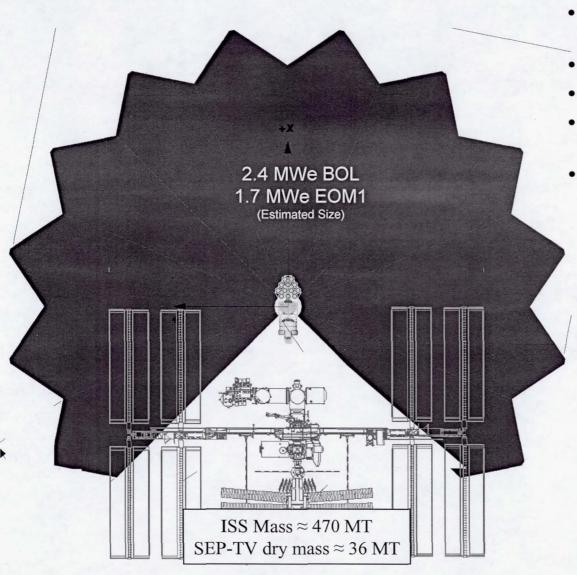
Key Attributes

- Crew of 4
- Global lunar access 3 day
- Lunar polar outpost 30 days
- ISS integral as a staging platform
- Lunar missions serve as "stepping stones" by providing an opportunity to test technology and operational concepts, reducing risk of future exploration endeavors
- Architecture can be bought "by the yard" resulting in increasing capabilities and operational experience
- Modest augmentation of commercial launch vehicles
- Common architecture elements for all Earth's Neighborhood missions

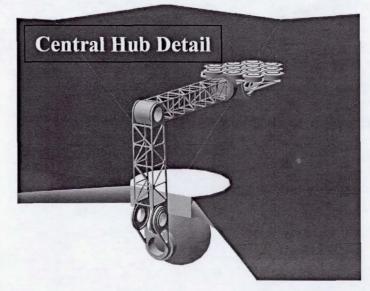


Solar Electric Propulsion Concept





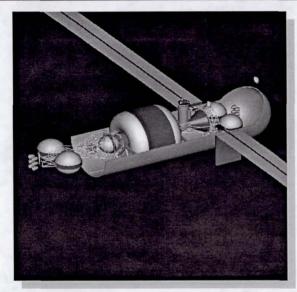
- Array sized to provide 1700 kW_e throughout first mission
- 14700 m² CuInS₂ array area
- 171 m span (wingtip-wingtip)
- 17 x 100 kW_e Hall Thruster Propulsion
- Articulated boom thruster gimballing





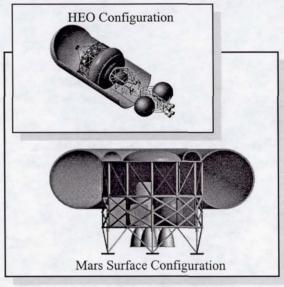
Mars Mission Vehicle Concepts





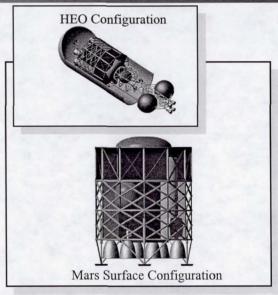
Mars Transit Vehicle

- Supports mission crew of six for up to 200-day transits to and from Mars
- Return propulsion stage integrated with transit system
- Provides return-to Earth abort capability for up to 30 hours post-TMI
- Total Vehicle Mass in High-Earth Orbit = 188 mt



Mars Surface Habitat

- Vehicle supports mission crew of six for up to 18 months on the surface of Mars
- Provides robust exploration and science capabilities
- Descent vehicle capable of landing 36,000 kg
- Total Vehicle Mass in High-Earth Orbit = 99 mt



Descent/Ascent Vehicle

- Transports six crew from Mars orbit to the surface and back to orbit
- Provides contingency abort-toorbit capability
- Supports six crew for 30-days
- Vehicle capable of utilizing locally produced propellants
- Total Vehicle Mass in High-Earth Orbit = 103 mt



Mars Mission Overview



